



CSAT 19 -20 June 2018
Australia/US SPD/Cold Spray Collaborations





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Introduction

- ❖ Update
 - Materials
 - Applications
- ❖ Australia/US Collaboration
 - History
 - Past Projects
 - Future Projects
 - Discussions/Questions




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Successful Application Summary

- ❖ Successful deposition of the following metal powders; Aluminium alloys (2000, 6000 and 7000 series), Titanium, Nickel, Steels (stainless, low carbon, mild, invar etc), Inconel, tantalum.
- ❖ >50 certified SPD repaired products released back into service, approx 10,000 accrued flight hours)
- ❖ Military helicopter transmission cost savings \$5M over four years. Significant down time reduction (TAT reduced by 75%)
- ❖ Components previously assessed as beyond economic repair can be returned to service
- ❖ Successful Commercial aerospace proof of concept applications Bombardier, Boeing , BAE, Messier Bugatti Dowty.

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Materials

| Material | Applications |
|---------------------|---|
| Al Alloy 6061 | Geometry restoration standard repairs, new proof of concept applications |
| Al Alloy 2024 | Geometry restoration , "structural applications" |
| Al Alloy 7075 | Geometry restoration, structural applications |
| Nickel/Nickel Blend | Geometry restoration, replacing electroplate nickel |
| 4340/M300 Steel | Geometry restoration, load bearing |
| Titanium | Geometry restoration, load bearing |
| Tungsten Carbide | Replace electroplate hard Chrome |
| Stainless Steel | Geometry restoration and corrosion protecting for mild steels |

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F/A18 Primary Manifold

Item received with severe shear damage on lock-ring bore wall. All serrations appear to have damaged or stripped during Rosan plug removal.



Mock-ups were created to mimic the damage removal. Port bore was machined to oversize dimensions: 14.20mm dia.



The finished trial has proved this to be an ideal SPD repair in both terms of deposition accessibility and machinability.

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NLG Wheel Hub

Item received with severe chatter marks and wear on the bearing surface of the hub. Largest damage depth recorded: 0.038 inch.



Bore coated with 2024 Al to a minimum thickness of 0.075 inch

SPD Coating machined to acceptable limits and refinished. Final coating thickness: ~0.020 inch



The SPD coating was tested for its ability to withstand shearing stress generated by the torque load. The coating can take 7 times more shearing (torque) load than the estimated design load at the bearing/bore interface due to interference fitting.



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SPD Wheel Hub Repair



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Oil Reservoir (T56)



Material: s/s 321;
 Thickness: 0.7mm?
 Defect: 35x5 mm, Depth: 0.178mm



- ✓ Cost of recovery including repair development < 15% of purchase price
- ✓ Turn around development and repair approx 8 weeks.

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SPD nickel Application to F/A18 part -Approved application

- Corrosion test: Salt spray passed, Rating 10
- Series of test passed: Bend, scraping, ring shear, etc
- Porosity: ~0%
- Hardness: >30 HRC
- Quality Satisfactory confirmed by both RUAG Australia and ASC



MLG Crank Assy
P/N 74A410555-series



MLG Crank Assy
P/N 74A410549-series



Planing Arm Assy
P/N 74A410538-series

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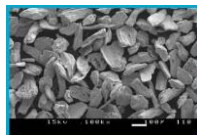
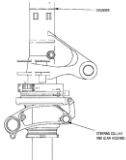
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NLG Steering Collars



Replacement of degraded Karon coating and repair of underlying corrosion with SPD lubricity metal deposition (Nickel graphite)



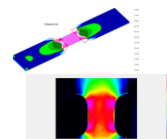
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Structural Integrity

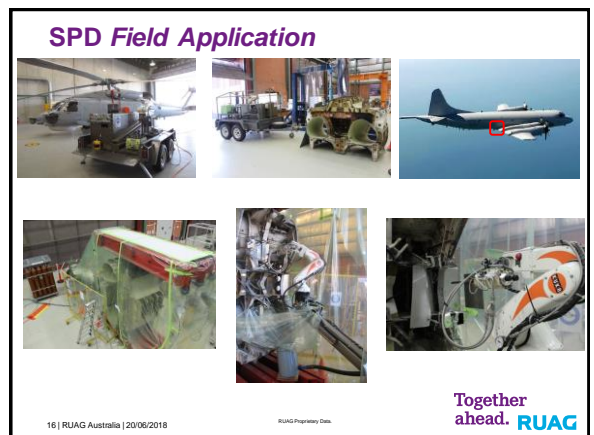
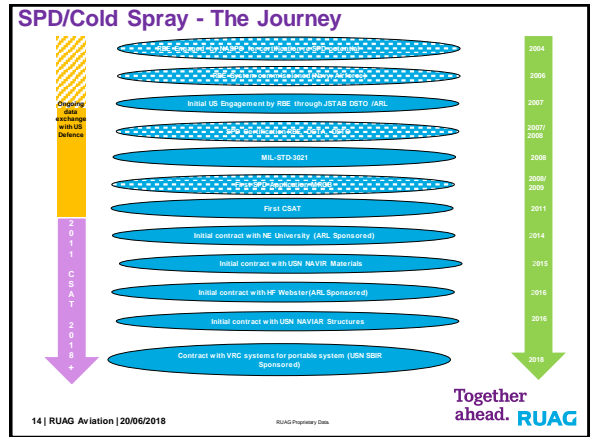
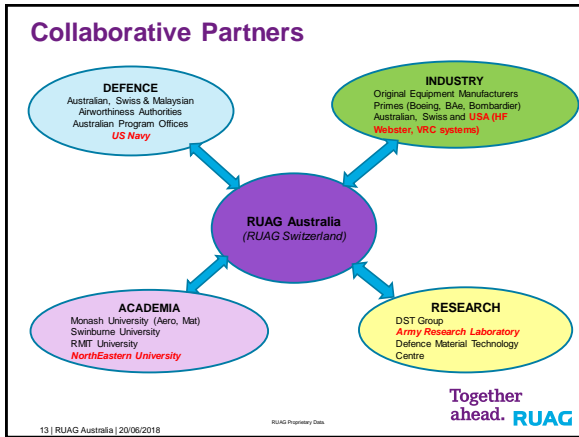
- Need to Quantify Cold Spray Performance on aircraft parts/structure
- Parts/structure be subject to both static and dynamic load conditions
- Measurements would be obtained from traditional strain gauging and thermography
- Current Opportunities
 - Swiss F/A-18 Door Frame structural enhancement to reduce peak stresses (underway)
 - Wing Structure dynamic tests (underway)
 - Fast Jet Full Fatigue Tests (Wing & Fuselage) (underway)







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Collaborative Programs With ARL

| Agency | Activity | Status | Outcome |
|---|---|----------------------|--|
|  | Development of High Ductile/Strength Cold Spray Aluminium Aerospace Alloys for Mobile Applications | Complete (2014-2015) | Replicating RUAG's Field Portable SPD Unit achievable spray parameters on our fixed system has demonstrated that additional pressure/temperature enhancements are required to achieve optimum deposition outcomes for 7075 Al alloy powder depositions. |
|  | Development of Advanced Cold Spray Prototype for Mobile Applications | Complete 2015-2016 | The assessment undertaken has identified a number of features on VRC Systems a Gen III High Pressure Hybrid Cold Spray System which have the potential to be incorporated into the FPSFDU to produce a highly effective "cold spray system" that could provide portable, robotically controlled deposition capability for numerous metal powders for a vast array of industries. |
|  | Develop a detailed specification for the integration of the provided VRC Gen III system into RUAG Australia's FPSFDU. | Complete 2017 | Gen III System provided by ARL Specification completed Layout Modelled confirmed |
|  | Optimization of Advanced Cold Spray Mobile System through | In work | Gen III system operated and particle velocity measured under different parameters to optimum parameters to align with fixed plant operations. Revised nozzle design generated. Ruggedisation requirements identified. |


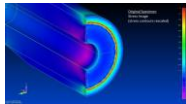
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Development of High Ductile/Strength Cold Spray Aluminium For Mobile Applications

1. Dense, well adhered low porosity coating for all three powder types can be achieved through parameter optimisation.
2. The deposition of all three powder types under optimised parameters will satisfy geometry restoration requirements.
3. Replicating RUAG's Field Portable SPD Unit achievable spray parameters on our fixed system has demonstrated that additional pressure/temperature enhancements are required to achieve optimum deposition outcomes for 7075 Al alloy powder depositions.
4. Modified axial tensile test coupons should be used to eliminate the possible impact of stress concentrations at the substrate coating edge.

| Stand-off distance (mm) | Temperature/Pressure (C/bar) | Label | Failure load (N) | Failure Stress (kN/m²) |
|-------------------------|------------------------------|-------|------------------|------------------------|
| 35-40 | 295/25 | 7A-1 | 954 | 185 |
| 35-40 | 425/38 | 7B-1 | 7,020 | 213 |
| 35-40 | 425/38 | 7B-2 | 4,317 | 151 |
| 50 | 425/38 | 7C-1 | 4,934 | 159 |
| 50 | 425/38 | 7C-2 | 6,091 | 185 |
| 75 | 425/38 | 7D-1 | 6,543 | 193 |
| 75 | 425/38 | 7D-2 | 7,323 | 186 |

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Detailed Specification (modelled)





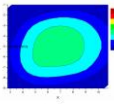
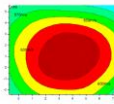
Hybrid Portable System

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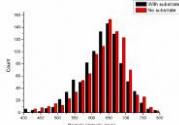
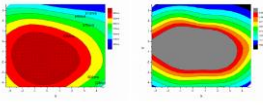
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Optimization of Advanced Cold Spray Mobile system

Velocity distribution in SPD jet sections at a stand-off of 35mm using VRC NZ2,0050 nozzle (left) and RUAG 75A,3089-2/-50nozzle (right).

3.2 Optimization 2
 Parameters: 600psi-440°C, 30SLM-4rpm

Particle velocity (left) and particle flow (right) distribution at the selected optimum condition 2

Particle velocity distribution comparison of SPD with/without substrate

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Collaborative Programs With USN



| Agency | Activity | Status | Outcome |
|--------------------|---|----------------------|--|
| NAVIAR Structures. | Restoration of Load Carrying Capacity of Wing Skins and Structure FEA and Coupons | Complete | The analyses and validation testing outlined in this project demonstrated that SPD repairs are viable options to restore integrity |
| NAVIAR Structures | On going structural integrity evaluation of SPD coating on aircraft structures | In work | Test Plan complete, Aircraft wing segments prepared |
| NAVIAR Materials | Trial SPD coatings on aircraft component | Complete | Quality and fatigue of coated components assessed independently No issues. |
| VRC | Acquisition of Mobile cold spray FPSPDU for trials and evaluation in 2019 | In work. Contact let | Ruggardistaion Specification finalised |

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Assessment of Wing Integrity with Upper Surface Corrosion Subject to Compressive Loading

Baseline Model

- A 500 mm (long) by 250 mm (wide) rib stiffened 7075-T6 aluminium alloy panel where the skin and the risers were both 4.3 mm thick.
- The panel contained 5 ribs
- The length of the panel was taken so as to approximate the distance between the H-clips that are attached to the risers.
- The risers were 30 mm deep and 50 mm apart.
- The skin thickness analysed was 2.1 mm.
- The unloaded edges of the panel were assumed to be simply supported (other cases were analysed)



Model with Corrosion removal

- The corrosion was taken to be 40 mm in diameter with a depth of 1.05 mm. (50% of the total thickness of the skin)
- Centre of Panel for maximum Buckling stress

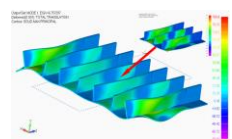
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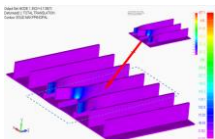
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On going structural integrity evaluation of SPD coating on aircraft structures

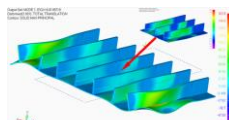
- Analysis was repeated for a 7 stiffener panel.



Buckling load (No SCC) - 946 kN



Buckling load (3 risers with SCC) - 510kN



Application of 3 thin 0.2 mm thick, 10 mm wide, and 110 mm long SPD doublers essentially **restored** both the buckling load and the buckling mode.

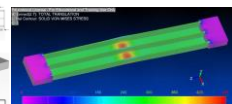
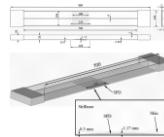
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On going structural integrity evaluation of SPD coating on aircraft structures

- The overall program objective is to produce quantifiable structural enhancement outcomes using SPD.
- This program is a one year program to quantify the contribution of al alloy 7075 SPD to the structural integrity of a repaired aircraft 7075-T6 framed. Measurements will be obtained from traditional strain gauging and thermography both under static and dynamic load conditions. The results will be compared to analytical predictions.



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Demonstration of the Portable CS System @ RUAG Rosebank (Australia)

Courtesy of USN 'In-Service Portable Dimensional Correction Repair for Aircraft' April 2015, OSD Correction Office Project W14NA07, NAVAR 4.3.4 & 4.3.3

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SPD Field Application

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USN Purchase Of Mobile Deposition Unit

Prime Contract: USN – VRC (Under existing SBIR No. N68335-17-C-0556:0 (sole source for 5 years)
 VRC Contract with RUAG – VRC/RUAG/NAVY SBIR N6833517C0556/050818 executed 28 May 2018
 Delivery: FEB 2019

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QUESTIONS ?????